2 PLACES

DATE OF COMPLETION: PARTICIPANTS: _____

DATE: 1/30/18 REVISION: 0

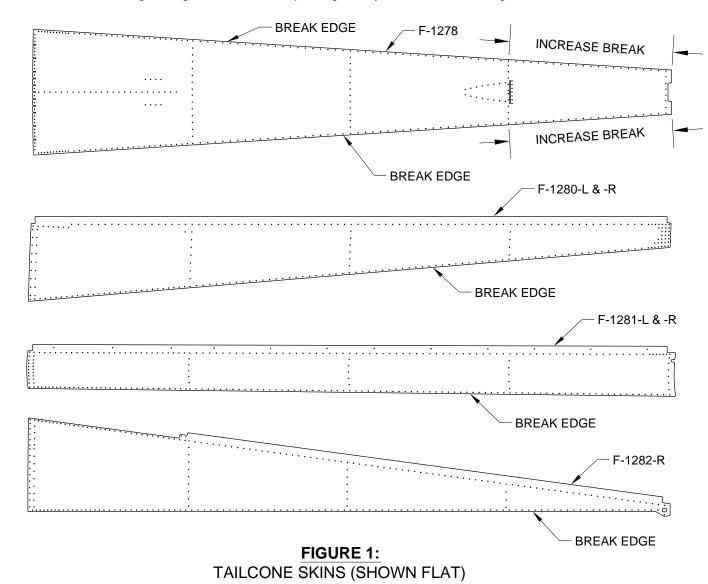
RV-12iS

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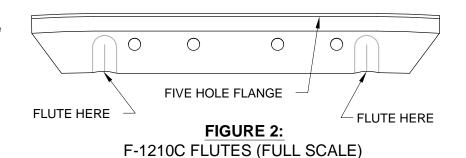
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Step 1: Identify the tailcone skins. The F-1278 Top Skin does not have a J-stiffener formed on either of the long edges. The F-01282-L-1 Bottom Left Skin has J-stiffeners formed on both of the long edges, while the F-1282-R Bottom Right Skin has only one. The remaining tailcone skins have a J-stiffener formed on one edge, and each left skin is mirrored by a right.

Step 2: Prepare the skins shown in Figure 1 for a lap joint by deburring and breaking (Section 5K) the edges. The edges to be lap joints are called out in Figure 1. Both edges of the F-1278 Top skin must break in the same direction. The remaining skins must break in the same direction as the J-stiffener. Begin with the F-1282-R Bottom Right skin, since it is least likely to be seen. When breaking the edges of the F-1278 Top Skin gradually increase the break angle toward the aft end.



Step 3: Flute the F-1210C Angle at the places called out in Figure 2. Each flute needs to 'hump' in the same direction as the five-hole flange. With the five-hole flange pointing up, align the holes in the four-hole flange of the angle to the full scale illustration in Figure 2. Increase the two flutes until the five-hole flange matches the curve shown in Figure 2.



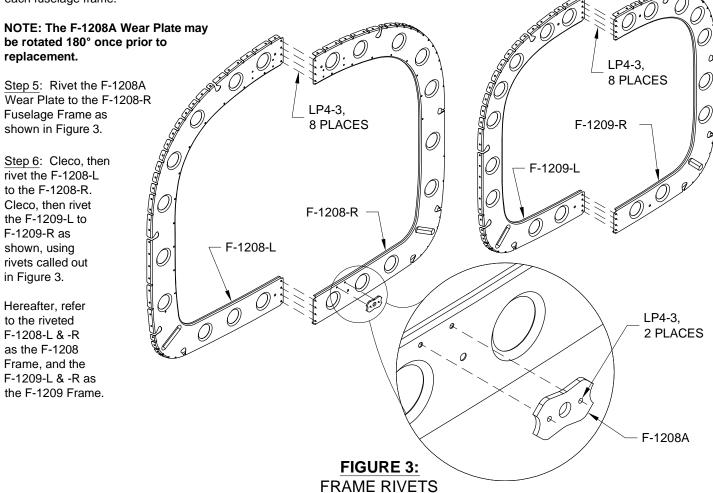
Step 4: Flute the F-1208-L & -R, F-1209-L & -R, and F-1210A-L & -R Fuselage Frames to align the flange holes. Flute the 'hump' of any humped flanges toward the web of each fuselage frame.

Step 5: Rivet the F-1208A Wear Plate to the F-1208-R Fuselage Frame as shown in Figure 3.

Step 6: Cleco, then rivet the F-1208-L to the F-1208-R. Cleco, then rivet the F-1209-L to F-1209-R as shown, using rivets called out in Figure 3.

replacement.

Hereafter, refer to the riveted F-1208-L & -R as the F-1208 Frame, and the F-1209-L & -R as the F-1209 Frame.



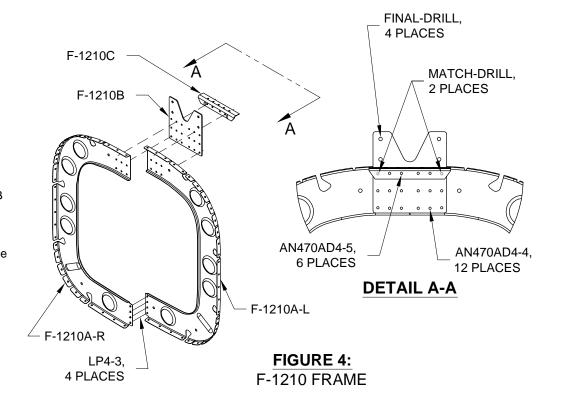
Step 7: Cleco the F-1210B Plate, and F-1210C Angle to the F-1210A-L & -R Fuselage Frames, as shown in Figure 4.

Step 8: Match-Drill #30 the two holes from the F-1210A-L & -R Fuselage Frames into the F-1210C Angle as called out in Figure 4. Disassemble and deburr.

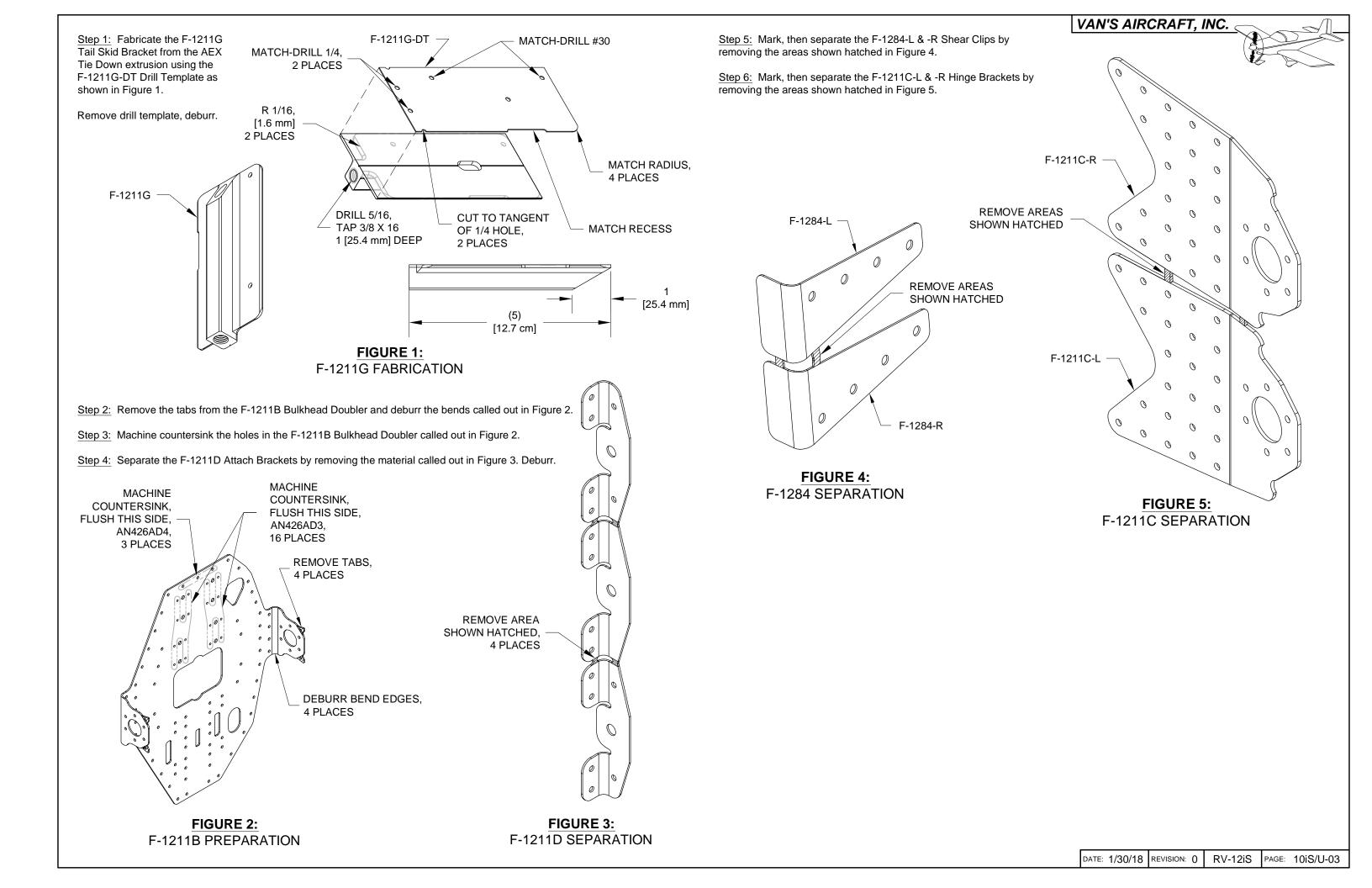
Step 9: Final-Drill #12 the F-1210B Plate at the holes called out in Figure 4.

Step 10: Rivet the F-1210A-L to the F-1210A-R Fuselage Frame, the F-1210B Plate, and the F-1210C Angle together using rivets called out in Figure 4.

Hereafter, refer to the riveted F-1210A-L & -R, F-1210B, and F-1210C as the F-1210 Frame.



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CAUTION: In Step 1 deburr only enough to locate the bushing.

Step 1: Rivet the F-1211D Attach
Brackets together per Figure 1. Deburr
the 1/4 in. holes per call-outs. Install the
bushing (oversize before press fit) as
shown. If necessary use a C-clamp with
a small socket opposite the bushing.

Step 2: Flute the F-1211A Fuselage Bulkhead as called out in Figure 2 only enough to reduce the slight pucker in the flange. Cracking may result from excessive force.

Step 3: Cleco the F-1211B Bulkhead Doubler to the F-1211A Fuselage Bulkhead as shown in Figure 2.

Step 4: Cleco the F-1211G Tail Skid Bracket to the F-1211A Fuselage Bulkhead as shown in Figure 2. Clamp the tail skid bracket in place.

Step 5: Match-Drill #30 the holes of the F-1211B Bulkhead Doubler into the F-1211G Tail Skid Bracket. Remove the tail skid bracket and deburr the holes.

<u>Step 6:</u> Final-Drill #12 the 3/16 holes called out in Figure 2. Disassemble, deburr, clear away chips, and re-cleco in place.

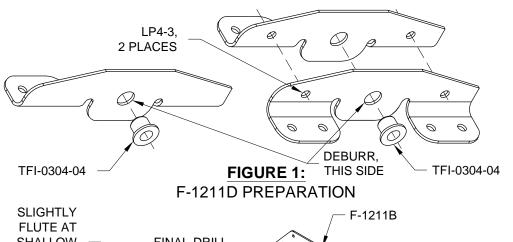
Step 7: Cleco the F-1211G Tail Skid Bracket and the F-1211D Attach Brackets (with bushing flange on the outboard side) to the F-1211A Fuselage Bulkhead and F-1211B Bulkhead Doubler.

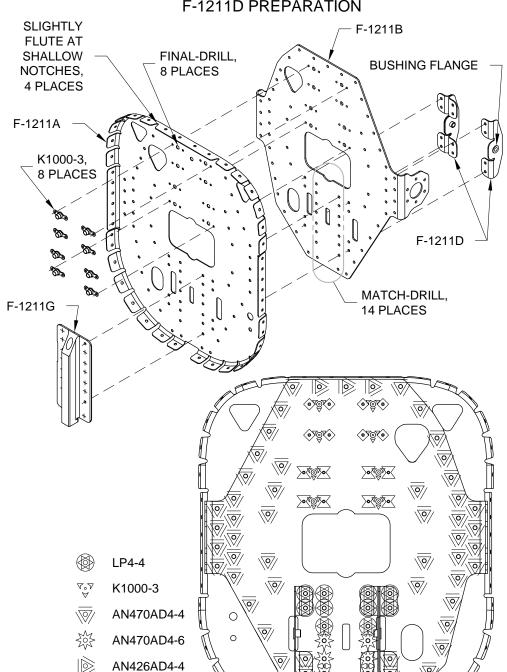
Step 8: Rivet the F-1211A Fuselage Bulkhead to the F-1211B Bulkhead Doubler, using the rivets indicated in Figure 2.

Step 9: Rivet the F-1211D Attach Brackets and F-1211G Tail Skid Bracket to the F-1211 Assembly using the rivets indicated in Figure 2.

Hereafter refer to the F-1211A Fuselage Bulkhead and the F-1211B Bulkhead Doubler as the F-1211 Assembly.

Step 10: Rivet the nutplates to the F-1211 Assembly as shown using hardware indicated in Figure 2.





F-1211 ASSEMBLY

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Step 11: Cleco the F-1211C-L & -R Hinge Brackets and the VA-146 Flange Bearings to the F-1211 Assembly as shown in Figure 3.

Step 12: Final-Drill #30 the holes common to the F-1211C-L & -R Hinge Brackets, the VA-146 Flange Bearings, and the F-1211 Assembly. Cleco each hole before drilling the next. Disassemble, deburr, then re-cleco in place.

Step 13: Rivet the F-1211C-L & -R Hinge Brackets and the VA-146 Flange Bearings to the F-1211 Assembly as shown in Figure 3.

Step 14: Insert the snap bushing into the F-1211 Assembly called out in Figure 3.

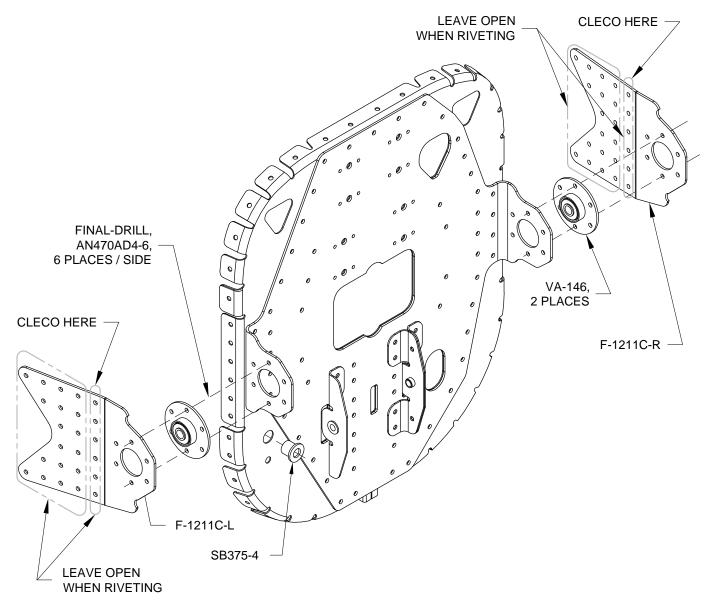
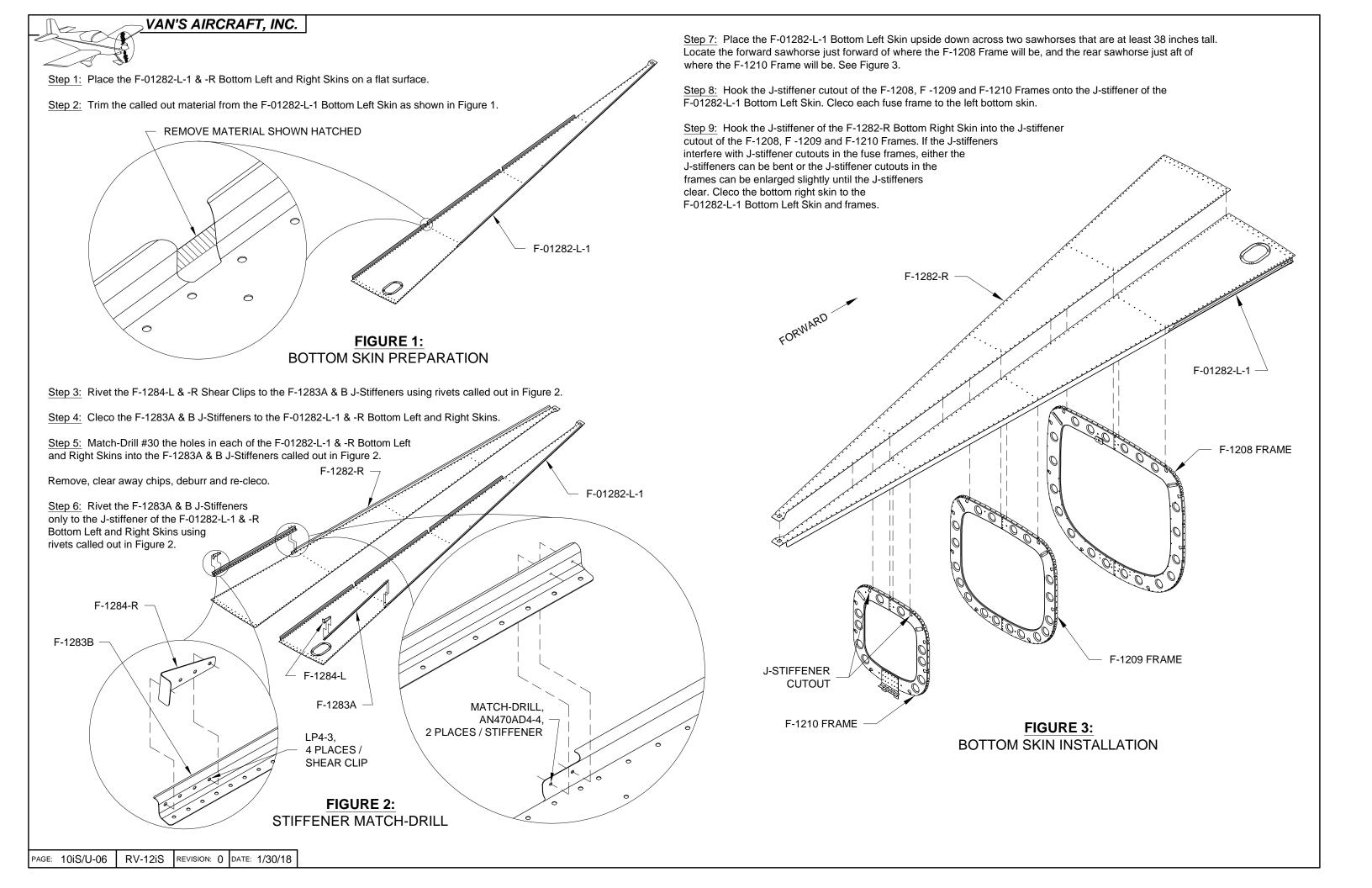
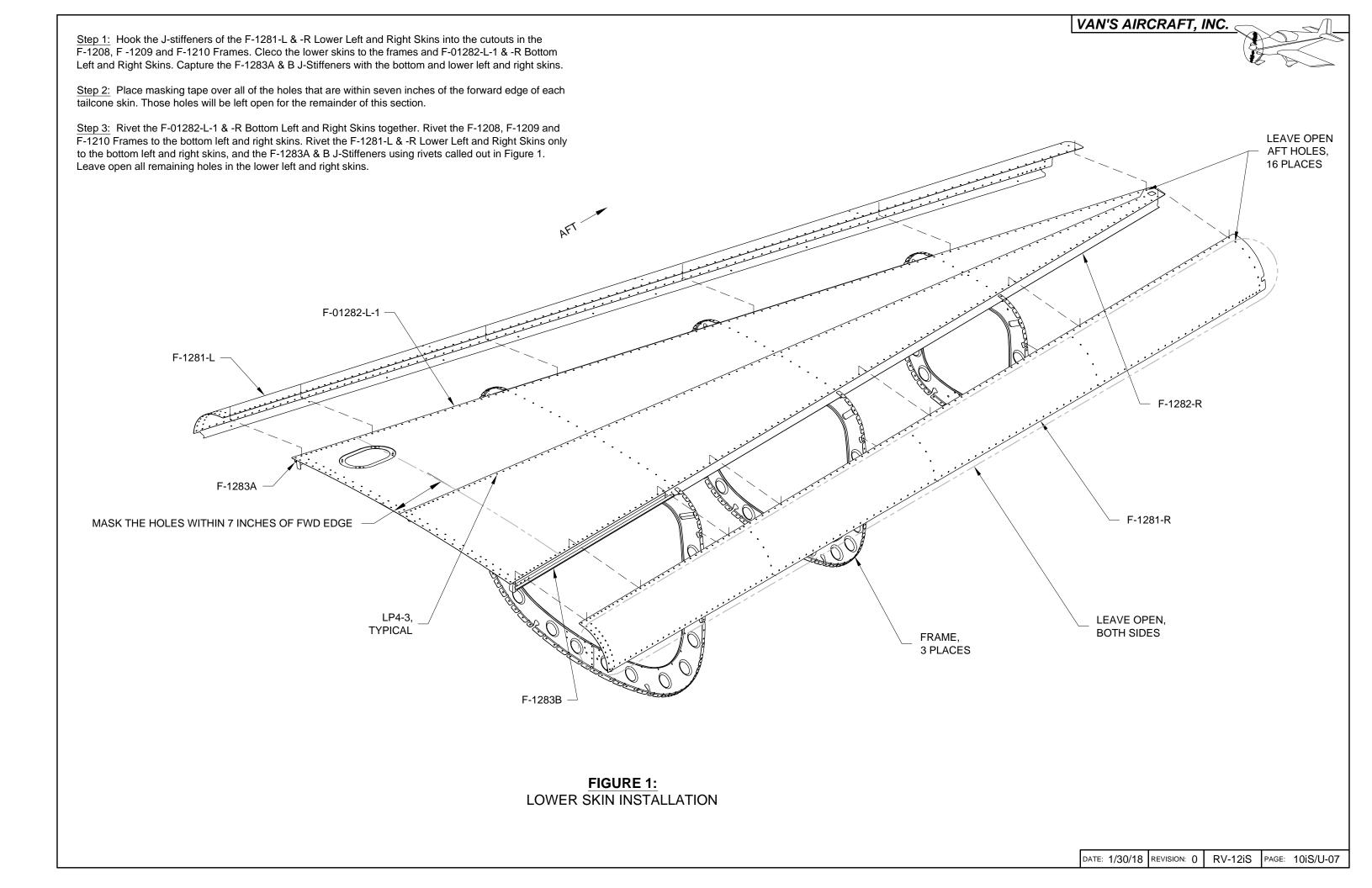
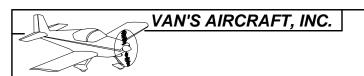


FIGURE 3:
BUSHINGS AND HINGE BRACKETS

VAN'S AIRCRAFT, INC. Step 1: With the Stabilator Assembly on a workbench, check the fit of the F-1211 Assembly hinges with the Step 2: Temporarily super glue the washers, determined in Step 1, to the Stabilator Assembly hinge brackets. Determine the best symmetric choice of washers to minimize lateral play outboard side of the HS-1213A Inboard Hinge Bracket and the inboard side of the HS-1213B Outboard Hinge and prevent binding throughout the stabilator travel. While determining the hinge hardware, use one washer Bracket. Be sparing with the super glue, contact only the washer and hinge bracket. Apply a minimal amount of between the bolt head and bracket, and use only one washer on each side of the bearing. The called out oil to the bolt shank, then locate the washers using the bolt called out in Figure 2. AN960-416 Washers may be substituted with AN960-416L Washers as necessary. Repeat this step for the remaining inboard and outboard hinge bracket. AN960-416L (OR -416), FROM STEP 1 STABILATOR AN960-416L (OR -416), ASSEMBLY FROM STEP 1 AN4-12A, 2 PLACES AN960-416L 2 PLACES HS-1213B HS-1213A AN4-12A FIGURE 2: WASHER PLACEMENT AN960-416L (OR -416 IF/AS REQ'D), 2 PLACES AN960-416L (OR -416 IF/AS REQ'D), 2 PLACES F-1211 ASSEMBLY FIGURE 1: STABILATOR HINGE WASHERS DATE: 1/30/18 REVISION: 0 RV-12iS PAGE: 10iS/U-05







Step 1: Turn the assembly over so the frames are pointing up. Support the assembly with two sawhorses, one at the F-1208 Frame and one at the F-1210 Frame.

<u>Step 2:</u> Cleco the F-1211 Assembly to the aft end of the existing structure as shown in Figure 1.

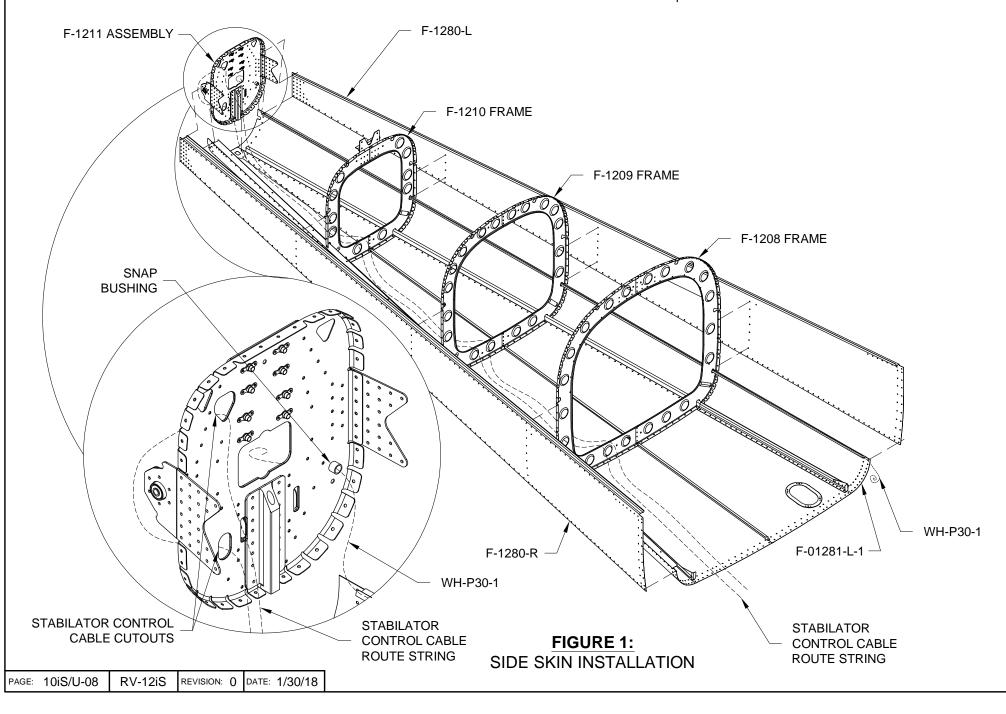
<u>Step 3:</u> Hook the J-stiffener of the F-1280-L & -R Left and Right Side Skins into the notches of the F-1208, F-1209, F-1210 Frames. Cleco the left and right side skins to the existing structure as shown in Figure 1.

Step 4: Route a string from the forward end of the existing structure, through the center of each frame, to the F-1211 Assembly. The string will go out through one of the stabilator control cable cutouts and back in through the other stabilator control cable cutout as shown in Figure 1. Return the end of the string to the forward end of the existing structure. Tape both ends of the string to the inside surface of the bottom skin.

NOTE: For the remaining steps in this section, leave the plastic tie wraps loose on the F-1208 Frame and the J-stiffener forward of the F-1208 Frame for additions later.

<u>Step 5:</u> Route the WH-P30-1 Trim Wires through the snap bushing in the F-1211 Assembly aligning the red band of heat shrink around the trim wires with the snap bushing. See Page 11-09, Figure 2.

Route the trim wires inside the J-stiffener 'hook' of the F-1281-L Lower Left Skin to the forward edge of the existing structure. At the holes provided in the J-stiffener aft of the F-1208 Fuse Frame secure the trim wires using plastic tie wraps. Coil the remainder of the trim wires and secure the coil to the inside of the lower left skin with tape.



NOTE: Fabricate F-00013 for Skyview ADAHRS installation only. Omit F-00013 for Garmin EFIS system. Download and read Section 42N before configuring the Static and Pitot lines if you are installing the Garmin EFIS System.

<u>Step 1:</u> Follow the 'Installing the Static Air System' instructions to install the left and right side static sources into the F-1280-L & -R Left and Right Side Skins in the locations called out in Figure 1.

Step 2: Cut two lengths of PT 1/8 CLR PLASTIC each piece 24 inches long to make two F-00012 Static Line Port - Tees.

<u>Step 3:</u> Install the end of one of the F-00012 Static Line Port - Tees over the right static source and install one end of the other static line port - tee over the left static source.

Seal the joint between the static line port-tees and the static source by applying a fillet of RTV silicone sealant as shown in Figure 2.

Route each static line port - tee upward and inboard along the inside flange of the F-1208 Frame as shown in Figure 1.

Secure the right side static line port - tee with plastic tie wraps through the holes along the flange of the frame. See Figure 1.

Install, but do not tighten, the plastic tie wraps on the left side. The left side tie wraps must be left loose so as to allow for other items to be routed along with the static line port - tee.

Step 4: Cut a length of PT 1/4 OD TUBE 15 inches long to make the F-00013 Static Line Tee - ADAHRS.

<u>Step 5:</u> Cut a length of PT 1/8 CLR PLASTIC 7/16 inches long to make the F-00014 Static Line Tee Upsize.

Step 6: Heat then slide the F-00014 Static Line Tee Upsize and F-00013 Static Line Tee - ADAHRS over the forward facing leg of the F PLASTIC TEE as shown in the exploded detail view of Figure 1.

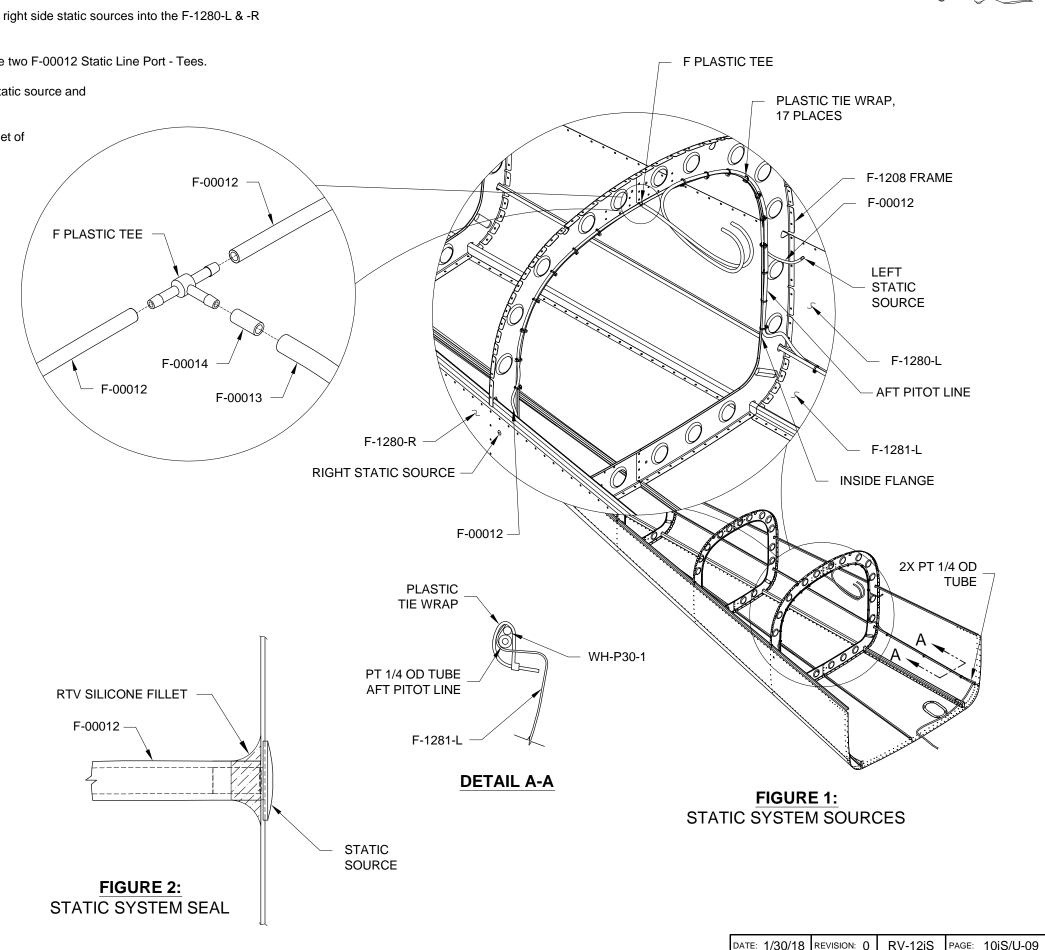
Heat then slide the two F-00012 Static Line Port - Tees onto the remaining legs of the F PLASTIC TEE as shown in the exploded detail view of Figure 1.

<u>Step 7:</u> The remaining length of PT 1/4 OD TUBE becomes the AFT PITOT LINE. Align one end of the Aft Pitot Line with the forward end of the F-00013 Static Line Tee - ADAHRS. See Figure 1.

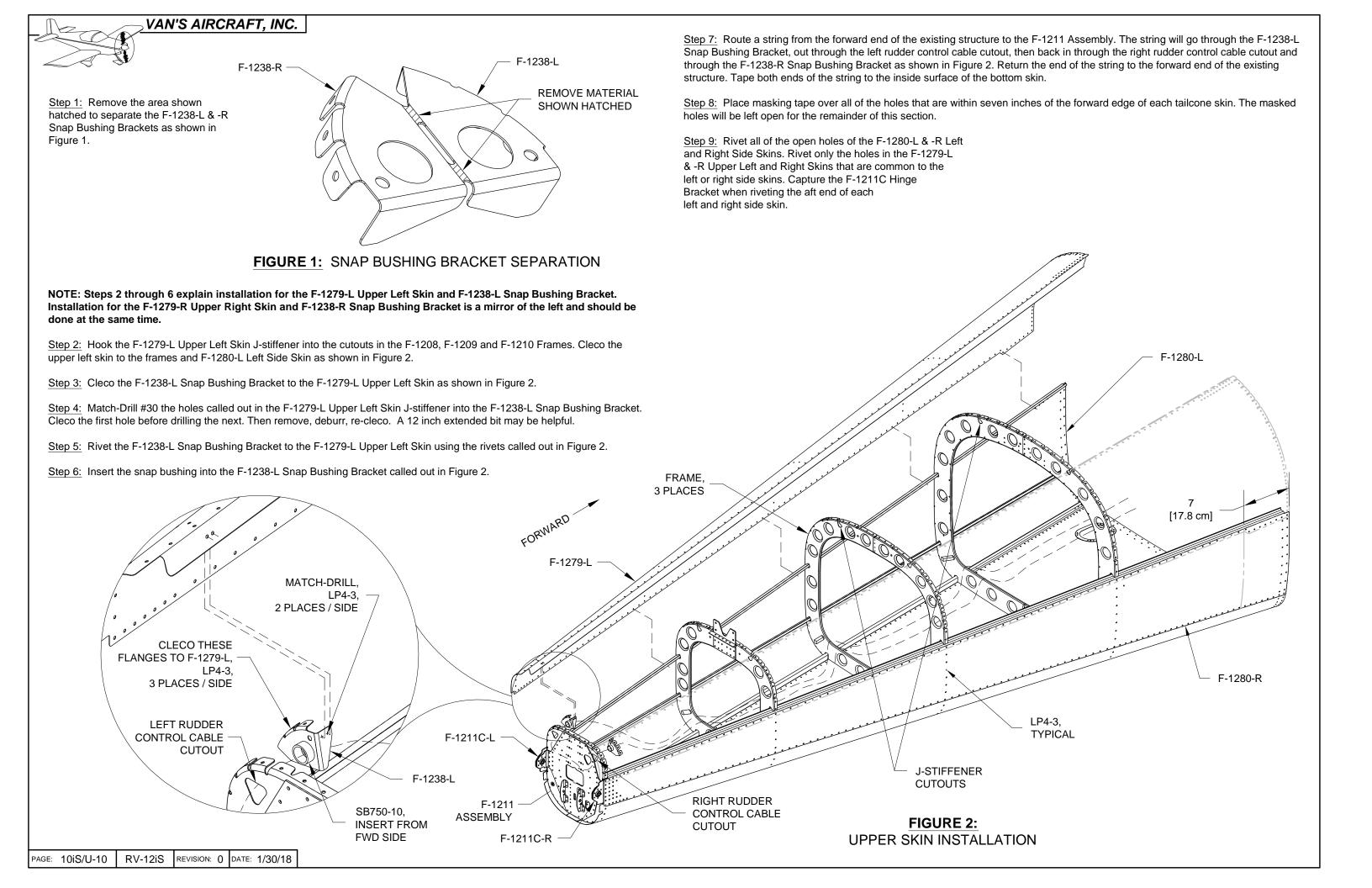
Route the Aft Pitot Line outboard and down the left side F-1208 Frame then to the J-stiffener of the F-1281-L Lower Left Skin as shown in Figure 1.

Route the Aft Pitot Line forward, running inside the J-stiffener 'hook' of the lower skin with the WH-P30-1, as shown in Detail A-A to the forward edge of the existing structure.

Coil the remainder of the Aft Pitot Line and secure the coils to the inside of the lower left skin with tape or string.



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Step 1: Final-Drill #19 the screw holes of the F-12106 Fwd Skin Rib per call-outs in Figure 1.

Step 2: Dimple #8 the screw holes of the F-12106 Fwd Skin Rib per call-outs in Figure 1.

<u>Step 3:</u> Dimple 3/32 the nutplate attach holes of the F-12106 Fwd Skin Rib per call-outs in Figure 1.

<u>Step 4:</u> Dimple 3/32 the nutplate attach holes of eight nutplates per call-outs in Figure 1.

Step 5: Rivet the dimpled nutplates to the F-12106 Fwd Skin Rib using hardware called out in Figure 1.

Step 6: Guide the F-1210B Plate through the slot in the F-1278 Top Skin. Cleco the Top Skin to the F-1210 Frame. Match-Drill #30 the two outboard-most holes in the F-1210C Angle using the Top Skin as a guide. Check the top skin for deformations caused by the F-1210C Angle at the locations called out in Figure 2. Remove the top skin. Deburr the angle. If deformations were present in the top skin file material from the angle to minimize deformations on the top skin. Cleco the top skin to the F-1208, F-1209 and F-1210 Frames and F-1279-L & -R Upper Left and Right Skins as shown in Figure 2.

Step 7: Rivet the F-12106 Fwd Skin Rib to the F-1278 Top Skin as shown in Figure 2.

Step 8: Cleco, then rivet the F-1283C J-Stiffener to the F-1278 Top Skin as shown in Figure 2, Detail A-A.

<u>Step 9:</u> Cleco, then rivet the F-00009-L & -R ADAHRS Brackets to the F-1278 Top Skin as shown in Figure 2, Detail A-A.

<u>Step 10:</u> Place masking tape over all of the remaining open holes that are within seven inches of the forward edge of the top skin.

Step 11: Rivet all of the remaining, unmasked open holes in the Tailcone Assembly using rivets called out in Figure 2.

